Seismic interferometry for earthquake-induced damage identification in historic masonry towers

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Abstract

The inherent vulnerability of masonry structures to seismic events makes Structural Health Monitoring of pivotal importance for the conservation of architectural heritage. In this regard, methods based on Operational Modal Analysis are becoming popular for damage identification. Nonetheless, these techniques may fail at detecting local damages with limited effects on the modal properties of the system. Recent studies report seismic interferometry to be a promising alternative for seismic damage identification of structures. This technique assesses the travelling times of propagating seismic waves between pairs of sensors, which are directly related to the local stiffness of the structure. Therefore, damage-induced degradation can be tracked through wave time delays. While some encouraging results have been reported on the application of acceleration-based seismic interferometry to reinforced-concrete structures, the number of works on masonry structures is far scarce. In this light, this paper is aimed at investigating the suitability of acceleration- and strain-based seismic interferometry for damage identification in historic masonry towers. To do so, an analytical layered Timoshenko beam model is devised for the wave propagation analysis of masonry towers under base motion. Parameter sensitivity analyses are first reported, with a special focus on the effects of dispersion upon system identification results. Secondly, a validation case study of a 41.6 m high masonry tower is presented. A realistic three-dimensional non-linear finite element model is built and subjected to seismic inputs causing increasing damage severities. The numerical results, used as pseudo-experimental data, demonstrate that it is possible to identify (detect, localize and quantify) earthquake-induced damages by wave propagation analysis of strain/acceleration records and inverse calibration of the proposed Timoshenko beam model. A particularly notable result is the possibility of detecting, localizing and, to some extent, quantifying earthquake-induced damage in a fully data-driven way by simply measuring wave travel times between pairs of sensors.

Keywords: Masonry towers, Wave propagation, Wave dispersion, Damage identification, Smart materials, Structural Health Monitoring

1 1. Introduction

There is a great awareness of the importance of conservation and safeguarding of heritage buildings as they 2 constitute vital assets with multiple positive socio-economic effects. According to the World Heritage List drawn 3 up by UNESCO, nearly half of the heritage sites are located in Europe, where Italy heads the list with 54 sites. It 4 is in Italy where the conservation of heritage buildings is specially critical due to its high seismicity, as evidenced 5 by recent severe events such as the Norcia M_w 6.5 earthquake occurred on October 30th 2016 [1]. In addition, 6 masonry structures, which represent a sizeable portion of the heritage assets, are characterized by low tensile 7 strength and intrinsic vulnerability to aging deterioration [2], whereby the assessment of their health condition is 8 of the utmost importance. In this context, Structural Health Monitoring (SHM) has proved to offer an efficient 9 solution, encompassing Non-Destructive Testing (NDT) and damage identification tools for the assessment of 10 the integrity of structures and preventive condition-based maintenance [3, 4]. In particular, vibration-based SHM 11 has received most attention in the realm of historic structures (just to name a few, see e.g. [5, 6, 7, 8, 9, 10]). 12 Output-only or Operational Modal Analysis (OMA) techniques exploit acceleration records to extract the modal 13 information of the system, namely natural frequencies, damping ratios, and mode shapes [11]. A large number of 14 15 output-only algorithms can be found in the literature, including among others Eigensystem Realization Algorithm (ERA) [12], Natural Excitation Technique (NExT) [13], Stochastic Subspace Identification (SSI) [14], Frequency 16 Domain Decomposition (FDD) [15], or Blind Source Separation (BSS) [16]. In this light, these techniques offer 17

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solutions for condition monitoring that are physically based on detecting damage-induced changes in the identified 18

modal features [17, 18, 19]. 19 While highly effective for the identification of damages affecting the overall stiffness of structures, OMA may 20 fail at detecting localized damages with limited effect on the modal properties of the system. As an alternative, 21 wave-propagation approaches conceive the seismic response of a structure as a superposition of waves propagating 22 through the structure, reflecting from its boundaries and interfering [20, 21, 22]. These techniques, also termed 23 seismic interferometry, exploit the seismic pulses or Impulse Response Functions (IRFs) generated by deconvo-24 lution of the recorded seismic response at different receivers in the structure [23, 24, 25]. The scattering and 25 attenuation of these deconvolved waveforms illustrate the propagation of shear waves through the structure, which 26 substantially depends upon the intrinsic characteristics of the structure, namely wave velocity, attenuation factor, 27 resonant frequencies, mode shapes, etc. Interestingly, unlike modal methods, wave propagation approaches show 28 no sensitivity to soil-structure interaction effects [23, 26, 27, 28]. Hence, the stiffness degradation produced by 29 damage leads to local delays in the wave propagation through the damaged part of the structure [29, 24, 25, 30]. 30 It is thus possible to devise a sensing network with a limited number of sensors capable of tracking wave delays, 31 32 or alternatively damage-induced effects on the local stiffness, for damage identification purposes. Another advantage of seismic interferometric approaches for damage identification compared to OMA-based techniques is that 33 they work with earthquake records, so that issues related to signal-to-noise ratios and temperature effects are kept 34 minimal. 35 Despite not being an entirely new approach, only a few publications in the literature have reported on wave 36 propagation methods for damage identification and mostly on Reinforced Concrete (RC) buildings. It is worth 37 noting the work by Trifunac et al. [31] who investigated the damage-induced effects on wave travel times in a 38 7-storey RC building in Van Nuys (California, US) during the 1994 Northridge M_w 6.4 earthquake. Their results 39 demonstrated that earthquake-induced damages can be estimated through increases in wavenumbers (i.e. slower 40 phase velocities) between pairs of sensors surrounding the damage. Another noteworthy contribution was done 41 by Todorovska and Trifunac [24] who studied the changes in wave travel times in a 6-story RC building in El 42 Centro (California, US). By deconvolving the recorded strong motion in three non-overlapping moving-windows 43 (before, during and after the largest registered amplitude response), their results reported good correlation be-44 tween the variations in the wave travel times and the observed earthquake-induced damages. Ebrahimian and 45 Todorovska [32, 33] developed a homogeneous and a layered Timoshenko beam model of high-rise buildings 46 for system identification based on wave propagation analysis of earthquake records. Through a non-linear fitting 47 of the developed analytical solution against the identified waveforms, their results demonstrated the importance 48 of bending deformation and rotary inertia in the deformation of high-rise buildings during seismic events. Very 49 good agreements with experimental data were reported for a 9-storey RC building in Pasadena and a 54-storey 50 steel-frame building in Los Angeles (California, US), and their results highlighted the contribution of bending 51 deformation to the dispersive response of buildings under seismic actions. Furthermore, some experiences can be 52 found in the literature on the extension of deconvolution seismic interferometry to long duration ambient noise 53 measurements. It is worth noting the work by Prieto et al. [34] who proposed a temporal averaging scheme 54 of deconvolved ambient vibration records divided into overlapping windows. Their approach was successfully 55 tested on a 17-story steel moment-frame building located at the University of California. Similarly, Nakata and 56 Snieder [35, 36] applied deconvolution interferometry to ambient vibration data recorded in an 8-storey build-57 ing in Japan. In that work, the wave velocities and amplitude decays were computed from the first upgoing and 58 downgoing waves, that is to say the first casual and acausal waveforms propagating for both positive and negative 59 times. More recently, Sun et al. [37] proposed a Bayesian probabilistic updating of building models with response 60

functions extracted from ambient noise measurements using seismic interferometry. Those authors demonstrated 61 the effectiveness of the proposed algorithm with a case study of a 21-storey RC building. On the whole, while 62 considerable effort has been devoted to the monitoring of RC buildings, the number of applications to masonry 63

structures is sorely lacking. 64

In the realm of historic masonry structures, the monitoring of static parameters, such as strains, tilts or dis-65 placements, offers an alternative to modal analysis for assessing the performance of structures and tracking load 66 paths changes in order to infer the presence of damages [3]. Depending on the desired outcomes, diverse ap-67 proaches and techniques are available. Very often, Linear Variable Displacement Transducers (LVDTs) are used 68 in masonry structures for the monitoring and tracking of crack amplitudes [38], while local vertical stresses and 69 elastic moduli can be monitored by flat-jacks [39]. Moreover, different sensing technologies can be used which 70 include, among others, laser scanning [40], Ground Penetrating Radar (GPR) [41], 3D Digital Image Correlation 71 (DIC) [42], sonic tests [43], and Fiber Bragg Grating (FBG) sensors [44]. More recently, advances in the fields 72 of Materials Science and Nanotechnology have enabled the development of innovative nano- or micro-modified 73 composites with multifunctional capabilities, offering a vast potential for SHM applications [45]. Particularly 74 promising in civil infrastructures are the self-sensing cement-based composites, often termed "smart concretes". 75

These materials are typically enriched with carbon-based fillers such as carbon black, carbon nanofibers, carbon 76 nanotubes or graphene [46]. These do not only fulfil an enhanced structural function, but more interestingly 77 they also exhibit self-sensing piezoresistive properties apt for being exploited in a condition-based maintenance 78 approach [47]. These novel composite materials outperform conventional sensing technologies in terms of archi-79 tectural invasiveness and long-term reliability, since it is the own structure which monitors its condition. Different 80 application lines of smart concretes can be found in the literature, comprising integral smart structures [48], em-81 bedded sensors [49], as well as smart skins [50]. In the realm of masonry structures, the concept of self-sensing 82 structural masonry, also termed "smart bricks", was first introduced by Downey et al. [51]. Those authors manu-83 factured burned clay bricks doped with titanium dioxide and experimentally characterized their strain self-sensing 84 capabilities. Their results demonstrated that it is possible to detect damage-induced variations in the load paths 85 through electrical resistivity measurements in the smart bricks and, therefore, conduct condition-based mainte-86 nance applications. On the whole, strain-based monitoring systems play a predominant role in the monitoring of 87 historic structures. Nevertheless, to the best of the authors' knowledge, the analysis of wave propagation on the 88 basis of strain-based monitoring systems remains unexplored. 89 In view of the aforementioned literature review, this paper is aimed at investigating the application of acceleration-90 and strain-based wave propagation analysis for damage identification in masonry towers under seismic actions. To 91 do so, an analytical model is devised considering a cantilever Timoshenko beam model of masonry towers excited 92 by base motion. This model considers both uniform and non-uniform layered Timoshenko beams, and represents 93 an extension of the models previously reported by Ebrahimian and Todorovska [32, 33]. Here, the pulse propaga-94 tion of shear waves is extracted from acceleration and normal strain transfer functions at different heights along 95 the structure, and governing dimensionless parameters are analysed for a range of masonry towers. The wave dispersion is derived from the uncoupled equation of motion of the beam, whereas pulse propagation is analysed 97 using impulse response functions. For damage identification purposes, a layered Timoshenko beam model is de-98 rived analytically using the propagator matrix approach. Detailed parametric analyses are first presented from 99 the perspective of structural identification, with special focus on the effects of dispersion on the identification of 100 travelling pulses. Secondly, a representative validation case study of a 41.6 m high isolated masonry tower under 101 ground acceleration is also presented. To this aim, a realistic three-dimensional non-linear Finite Element Model 102 (FEM) is built and subjected to different earthquake loadings with increasing peak ground accelerations to generate 103 pseudo-experimental data. Wave propagation analyses are conducted by the post-processing of strain/acceleration 104 records at different heights of the FEM, and damage identification is performed by model updating of the proposed 105 Timoshenko beam model. Since only seismic excitations are considered, noise effects are disregarded in this work 106 without loss of generality due to the large signal-to-noise ratios typically present in such cases. 107 The remaining of this paper is organised as follows. Section 2 introduces the concept of seismic interferometry 108 through acceleration/strain sensors for earthquake-induced damage identification in masonry towers. Section 3 109

¹¹⁰ overviews the formulation of the wave propagation problem in a Timoshenko beam model excited by base motion.

Section 4 presents the numerical results and discussion and, finally, Section 5 concludes this work.

112 2. General concept of deconvolution seismic interferometry

Seismic interferometry is a technique originally proposed in Geophysics [52] to extract the seismic wave 113 propagation characteristics of a structure under seismic actions. In essence, this technique is aimed at constructing 114 a Green's function describing the wave propagation between a set of receivers (e.g. geophones, hydrophones, or 115 accelerometers) distributed along the monitored structure [53, 54]. The data processing of the recorded signals can 116 be based on cross-correlation, cross-coherence or deconvolution [55]. In particular, deconvolution interferometry 117 has been reported to be well-suited for the monitoring of mono-dimensional structures such as buildings [29, 24, 118 25, 30]. It is important to note that this technique differs from NDT of materials using ultrasonic waves. Since such 119 waves require the use of generators and are quickly attenuated, their applicability is limited to the identification 120 of defects in local members [56]. Conversely, seismic waves exhibit larger wavelengths of around 5-500 m and 121 experience little attenuation, thereby seismic interferometry can be used to characterize a large-scale building 122 without any actuator. Hence, given that the wave travel times solely depend on the intrinsic characteristics of the 123 structure, the appearance of damages can be tracked as wave delays between pairs of sensors. 124 In this paper, let us consider a masonry tower equipped with an array of sensors monitoring its response u(z, t)125

¹²⁵ In this paper, let us consider a masonry lower equipped with an array of sensors monitoring its response u(z, t)¹²⁶ along the height $0 \le z \le H$, where t is the time variable and H the total height of the tower. The deconvolution ¹²⁷ interferometry technique allows getting an insight into the propagation of waves between two arbitrary sensors, ¹²⁸ considering one sensor at level z_{ref} as reference input signal $u(z_{ref}, t)$ and the other at level z as output signal ¹²⁹ u(z, t). Assuming the tower as a linear time-invariant system, the reference and output signals are related in the

time domain t as [24]:

$$u(z,t) = u(z_{ref},t) * h(z, z_{ref},t) = \int_0^t u(z_{ref},\tau) h(z, z_{ref},t-\tau) d\tau,$$
(1)

or, alternatively, in the frequency domain ω as:

$$\widehat{u}(z,\omega) = \widehat{u}(z_{ref},\omega)\widehat{h}(z,z_{ref},\omega),$$
(2)

where * indicates convolution, and a hat indicates Fourier transform. Functions $h(z, z_{ref}, \omega)$ and $h(z, z_{ref}, t)$ denote 132 the transfer function (TF) and the impulse response function (IRF) between the output signal u(z, t) and the input 133 signal $u(z_{ref}, t)$, respectively. Note that, in general, the signals of the sensors, u(z, t) and $u(z_{ref}, t)$, may correspond 134 to different physical measurements such as acceleration, displacement or strain. Th IRFs physically relate the 135 responses of the system at different levels z to a virtual Dirac Delta impulse $\delta(t)$ at level z_{ref} . In other words, these 136 functions represent the Green's functions of the system at different heights of the structure and characterize the 137 propagation of an input pulse, applied at z_{ref} , between the receivers deployed in the structure [55]. Furthermore, 138 the IRFs can be computed from any recorded response (e.g. displacements, velocities, accelerations or strains), 139 by taking the inverse Fourier transform of the corresponding TFs as follows: 140

$$h(z, z_{ref}, t) = \mathscr{F}^{-1}\left\{\widehat{h}(z, z_{ref}, \omega)\right\} = \mathscr{F}^{-1}\left\{\frac{\widehat{u}(z, \omega)}{\widehat{u}(z_{ref}, \omega)}\right\},\tag{3}$$

where \mathscr{F}^{-1} denotes the inverse Fourier transform. The travelling times of the shear waves propagating in the structure can be computed using the IRFs and, subsequently, damages can be inferred from wave delays. It is important to note that the response at any level can be used as the reference, thus defining a virtual source. In the case of structures under seismic actions, the IRFs are commonly obtained by considering a virtual source either at the base or the roof. In the latter case, the input source does not coincide with the physical source, i.e. the base acceleration. Moreover, although acceleration records are typically used to describe the propagation of seismic waves, other physical measurements can be also used. In this work, the possibility of using strain-based waves is also invarting the structure of the st

148 also investigated.



Figure 1: Schematic of a possible SHM system for a masonry tower for damage detection and localization using strain sensors and deconvolution seismic interferometry.

As an illustrative example, Fig. 1 sketches a possible SHM system for damage detection and localization in a masonry tower using strain sensors and deconvolution seismic interferometry. In this case, four strain sensors (e.g. LVDTs, dynamic strain gauges or smart bricks) are deployed along the height of the tower and, therefore, the structure can be conceived as an ideal 4-layered medium. Note that there is no need to install a strain sensor

at the roof level because of the strain-free condition at that location. In order to characterize the seismic input 153 and conduct wave propagation analyses considering virtual sources both at the base and at the roof levels, at least 154 two accelerometers must be installed at these reference levels. Therefore, the IRFs are computed by deconvolving 155 the recorded strain signals with respect to the acceleration at the reference level. In this regard, Fig. 1 also shows 156 a hypothetical representation of the strain IRFs considering a virtual source at the roof level. Each section of 157 the tower delimited by pairs of consecutive sensors can be characterized by a value of shear wave velocity v_i . 158 Considering the approximate ray theory, that is to say, ray paths obeying the Snell's law, it is possible to identify 159 the ray paths along the tower as those marked in red in the example of Fig. 1. Hence, the shear wave velocity 160 in the *i*-th layer of height l_i can be computed by the identified wave travel time τ_i as $v_i = l_i / \tau_i$. Moreover, given 161 that the velocity of the pulse v_i is directly related to the local stiffness of the tower, damage-induced stiffness 162 reductions in the *i*-th layer can be detected by increases in the wave travel time τ_i . Note that the spatial resolution 163 of the damage localization is limited by the number of sensors, being two (base and roof) the minimum number 164 that is needed to detect damages in the structure. While the deconvolution interferometric approach can be fully 165 data-driven, theoretical wave propagation models are valuable tools to correctly identify the ray paths and interpret 166 the waveforms. For this purpose, a layered Timoshenko beam model for wave propagation analysis in masonry 167 towers is proposed hereafter. 168

3. Theoretical formulation

In this section, the non-uniform viscoelastic Timoshenko beam model proposed by Ebrahimian and Todorvorska [32, 33] is overviewed, and it is extended to account for the propagation of strain waves. Specifically, Section 3.1 first outlines the theoretical formulation of wave dispersion in uniform Timoshenko beams. Secondly, Section 3.2 extends this formulation with a propagator matrix approach for multi-layered Timoshenko beams and, finally, Section 3.3 datails the calculation of acceleration and strain IPEs.

finally, Section 3.3 details the calculation of acceleration and strain IRFs.

175 3.1. Visco-elastic Timoshenko beam model

Firstly, the masonry tower is modelled as a cantilever uniform visco-elastic Timoshenko Beam (TB) subjected to seismic base motion u_g as sketched in Fig. 2. The beam has a cross-section *A*, second moment of inertia *I*, shear correction factor κ , width *W*, and height *H*. The material is defined as elastic isotropic with Young's modulus *E*, shear modulus *G*, and mass density ρ . The longitudinal and shear wave velocities in the material are defined as $c_L = \sqrt{E/\rho}$ and $c_S = \sqrt{G/\rho}$, respectively [32].



Figure 2: Uniform cantilever Timoshenko beam model.

The Timoshenko beam theory takes into account both shear deformation and rotational bending effects. To this aim, this theory distinguishes $\theta(z, t)$ and $\gamma(z, t)$ as the angles representing the rotations of an infinitesimal beam element located at height z due to bending and shear effects, respectively. Let us denote $u(z, t) = u_g(t) + u_r(z, t)$ the absolute horizontal displacement of the centre of gravity of the element with respect to the origin, where $u_r(z, t)$ stands for relative displacement. In addition, let us define v(x, z, t) as the longitudinal displacement of the beam.

Assuming small deformations, one can write the following kinematic conditions:

$$\frac{\partial u}{\partial z} = \theta + \gamma, \quad v = -x\,\theta,$$
(4)

and the shear forces V and bending moments M can be expressed in terms of rotations as:

$$V = \kappa A G \gamma, \quad M = -E I \frac{\partial \theta}{\partial z}.$$
 (5)

¹⁸⁸ Damping is included in this work by using the Kelvin-Voigt material damping model. According to this ¹⁸⁹ model, dissipative damping forces are taken into account by defining bending and shear stresses, σ_x and τ_{xy} , as ¹⁹⁰ linear functions of the strain velocity through two constants c_{σ} and c_{τ} , what leads to the following stress-strain ¹⁹¹ relations:

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$$\sigma_x = \sigma_x^e + \sigma_x^d = E\varepsilon_x + c_\sigma \frac{\partial \varepsilon_x}{\partial t},\tag{6}$$

$$\tau_{xy} = \tau_{xy}^e + \tau_{xy}^d = G\gamma_{xy} + c_\tau \frac{\partial \gamma_{xy}}{\partial t},\tag{7}$$

where superscripts "e" and "d" denote elastic and damping stresses, respectively. If we further decompose the damping constants c_{σ} and c_{τ} as $c_{\sigma} = \mu_{\sigma} E$ and $c_{\tau} = \mu_{\tau} G$, and we assume the same viscosity constant for both types of deformation, $\mu_b = \mu_s = \mu$, the Kelvin-Voigt damping model can be readily introduced by replacing *E* and *G* by $E [1 + \mu (\partial/\partial t)]$ and $G [1 + \mu (\partial/\partial t)]$, respectively.

The equations of motion of the TB can be obtained by applying the Hamilton's Principle to the Lagrangian \mathscr{L} and the Rayleigh dissipation function \mathscr{R} of the system:

$$\delta \int (\mathscr{L} - \mathscr{R}) \, \mathrm{d}t = \delta \int (T - U - \mathscr{R}) \, \mathrm{d}t = 0, \tag{8}$$

with T being the total kinematic energy of the beam (including the rotary inertia effect), and U the potential energy

due to bending and shear deformations. After some manipulations, Eq. (8) results in the following coupled system of equations:

$$\rho A \ddot{u} - \kappa G A \left(1 - \mu \frac{\partial}{\partial t} \right) (u^{\prime\prime} - \theta^{\prime}) = 0, \tag{9}$$

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$$\rho I\ddot{\theta} - EI\left(1 - \mu \frac{\partial}{\partial t}\right)\theta'' - \kappa GA\left(1 - \mu \frac{\partial}{\partial t}\right)(u' - \theta) = 0, \tag{10}$$

where dots and commas stand for time and spatial derivatives, respectively. If we denote $D = [u, \theta]^T$, Eqs. (9) and (10) can be rewritten in a decoupled system of fourth-order differential equations as:

$$EID^{\prime\prime\prime\prime\prime} + \rho A\ddot{D} + \frac{\rho^2}{\kappa G}\ddot{D} - \rho I \left(1 + \frac{E}{\kappa G}\right)\ddot{D}^{\prime\prime} + \mu GI\dot{D}^{\prime\prime\prime\prime} - \frac{\rho\mu I}{\kappa}\ddot{D}^{\prime\prime} + \mu \left(-\rho I\ddot{D}^{\prime\prime} + EI\dot{D}^{\prime\prime\prime\prime} + \mu GI\ddot{D}^{\prime\prime\prime\prime} + \rho A\ddot{D}\right) = 0, (11)$$

²⁰⁵ or in a more compact way as:

$$c_L^2 c_S^2 \kappa \left(1 + \mu \frac{\partial}{\partial t}\right)^2 \frac{\partial^4 D}{\partial z^4} - \left(c_L^2 + \kappa c_S^2\right) \left(1 + \mu \frac{\partial}{\partial t}\right) \frac{\partial^4 D}{\partial z^2 \partial t^2} + \frac{\kappa c_S^2}{rg^2} \left(1 + \mu \frac{\partial}{\partial t}\right) \frac{\partial^2 D}{\partial t^2} + \frac{\partial^4 D}{\partial t^4} = 0, \tag{12}$$

with r_g being the radius of gyration $r_g = \sqrt{I/A}$. The solutions of Eq. (12) satisfy the following boundary conditions:

$$\theta(0,t) = 0, u(0,t) = u_g(t) \text{ at } z = 0,$$
 (13)

$$V(H,t) = M(H,t) = 0$$
 at $z = H$. (14)

Assuming harmonic excitations, the solutions for transverse displacements and rotations can be defined as one-dimensional waves as $u(z,t) = e^{i(kz-\omega t)} = U(z)e^{-i\omega t}$ and $\theta(z,t) = e^{i(lz-\omega t)} = \Theta(z)e^{-i\omega t}$, respectively, where ω stands for angular frequency and i is the imaginary unit. Upon substitution of these solutions into Eq. (12), the equation for transverse displacements *u* can be rewritten as:

$$c_L^2 c_S^2 \kappa (1 - i\omega\mu)^2 k^4 - (c_L^2 + \kappa c_S^2) (1 - i\omega\mu) k^2 \omega^2 - \omega^2 \frac{\kappa c_S^2}{rg^2} (1 - i\omega\mu) + \omega^4 = 0.$$
(15)

Note that Eq. (15) establishes a relationship between the wavenumbers k and the frequency ω , also called a dispersion relation. Thence, the velocities of the waves are functions of the frequency. More specifically, the travelling waveform can be defined by the phase and group velocities as $c^p = \omega/k$ and $c^g = \partial \omega/\partial k$, respectively. Furthermore, note that the dispersion relation for rotations is identical to Eq. (15) and, therefore, so are their wavenumbers (l = k), phase and group velocities. In order to non-dimensionalize the equations for further analysis, a dimensionless frequency, Ω , a moduli ratio, R, and a dimensionless damping constant, M, are introduced as follows [32]:

$$\Omega = \frac{\omega r_g}{c_S}, \quad R = \frac{G}{E} = \frac{c_S^2}{c_L^2}, \quad M = \frac{\mu c_S}{r_g}.$$
(16)

In terms of these dimensionless parameters, the four roots of Eq. (15) yield the following dimensionless wavenumbers:

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$$K_{1,2} = \pm k_1 r_g = \pm \sqrt{\left(\frac{1}{\alpha}\right) \left(\frac{1}{\kappa} + R\right)} + \sqrt{\left(\frac{1}{\alpha^2}\right) \left(\frac{1}{\kappa} - R\right)^2 + \frac{4R}{\alpha \Omega^2}},\tag{17}$$

$$K_{3,4} = \pm k_2 r_g = \pm \sqrt{\left(\frac{1}{\alpha}\right) \left(\frac{1}{\kappa} + R\right)} - \sqrt{\left(\frac{1}{\alpha^2}\right) \left(\frac{1}{\kappa} - R\right)^2 + \frac{4R}{\alpha\Omega^2}},\tag{18}$$

with $\alpha = 1 - i\omega M$. Hence, the solutions of the lateral displacements U(z) and rotations $\Theta(z)$ in the frequency domain read:

$$U(z) = C_1 e^{ik_1 z} + C_2 e^{-ik_1 z} + C_3 e^{ik_2 z} + C_4 e^{-ik_2 z},$$
(19)

$$\Theta(z) = B_1 e^{ik_1 z} + B_2 e^{-ik_1 z} + B_3 e^{ik_2 z} + B_4 e^{-ik_2 z},$$
(20)

where C_i , B_i , i = 1, ..., 4 are constants determined by the boundary conditions in Eqs. (13) and (14). Substitution of Eqs. (19) and (20) into the coupled differential equations in Eqs. (9) and (10) leads to the following relations:

$$\frac{B1}{C1} = -\frac{B2}{C2} = -\frac{i}{rg} \left(\frac{\Omega^2}{K_1 \kappa \alpha} - K_1 \right),\tag{21}$$

$$\frac{B3}{C3} = -\frac{B4}{C4} = -\frac{\mathrm{i}}{rg} \left(\frac{\Omega^2}{K_2 \kappa \alpha} - K_2 \right),\tag{22}$$

and the boundary conditions in Eqs. (13) and (14) imply:

$$\begin{bmatrix} C_1 \\ C_2 \\ C_3 \\ C_4 \end{bmatrix} = \mathbf{A}^{-1} \begin{bmatrix} u_g(t) \\ 0 \\ 0 \\ 0 \end{bmatrix},$$
 (23)

230 with:

$$\mathbf{A} = \begin{bmatrix} 1 & 1 & 1 & 1 \\ -\left(\frac{\Omega^2}{\alpha\kappa K_1} - K_1\right) & \left(\frac{\Omega^2}{\alpha\kappa K_1} - K_1\right) & -\left(\frac{\Omega^2}{\alpha\kappa K_2} - K_2\right) & \left(\frac{\Omega^2}{\alpha\kappa K_2} - K_2\right) \\ \left(\frac{\Omega^2}{\alpha\kappa} - K_1^2\right) e^{iK_1(H/rg)} & \left(\frac{\Omega^2}{\alpha\kappa} - K_1^2\right) e^{-iK_1(H/rg)} & \left(\frac{\Omega^2}{\alpha\kappa} - K_2^2\right) e^{iK_2(H/rg)} & \left(\frac{\Omega^2}{\alpha\kappa} - K_2^2\right) e^{-iK_2(H/rg)} \\ -(1/K_1) e^{iK_1(H/rg)} & (1/K_1) e^{-iK_1(H/rg)} & -(1/K_2) e^{iK_2(H/rg)} & (1/K_2) e^{-iK_2(H/rg)} \end{bmatrix}.$$
(24)

Finally, the bending-induced normal strain at a beam depth x can be directly obtained by the kinematic condition in Eq. (4) as $\varepsilon(x, z, t) = \frac{\partial v}{\partial z} = -x \frac{\partial \theta(z, t)}{\partial z} = \Upsilon(z) e^{i\omega t}$. The strain in the frequency domain $\Upsilon(z)$ is extracted from Eq. (20) as:

$$\Upsilon(z) = -ik_1 B_1 e^{ik_1 z} + ik_1 B_2 e^{-ik_1 z} - ik_2 B_3 e^{ik_2 z} + ik_4 B_4 e^{-ik_2 z},$$
(25)

whereby it is concluded that the velocities of the waveforms arising from the strain field are identical to those obtained by monitoring lateral displacements in Eq. (19) or, alternatively, velocities or accelerations. Let us remark that, unlike acceleration-based wave propagation approaches, the position of strain transducers must be adequately tailored to maximize the bending-induced strains (typically x = W/2).

238 3.2. Visco-elastic layered Timoshenko beam model

In order to include non-uniform stiffness distributions and damage-induced variations, the tower is herein 239 modelled as a visco-elastic TB with piecewise continuous properties as shown in Fig. 3. In particular, n layers 240 are numbered from bottom to top, and are characterized by their height $l^i = z^i - z^{i-1}$, longitudinal and shear 241 wave velocities c_L^i and c_S^i , mass density ρ^i , and viscosity constants μ^i , i = 1, ..., n. It is important to note that 242 such a modelling framework allows the representation of damage in the shape of local reductions in the layers' 243 Young's modulus and/or shear modulus or, alternatively, local increases in the wavenumbers of the propagating 244 pulses in virtue of Eqs. (17) and (18). Similar assumptions are common in damage identification techniques via 245 computational model updating, being suitable for early-stage damage where the structure can be hypothesized to 246 remain elastic. The layering of the model allows the definition of multiple flaws located at different heights of the 247 tower, being the number of sensors deployed in the structure the only limiting factor in the spatial resolution of 248 the damage localization. Moreover, the inverse calibration of this model makes it possible to quantify damage in 249 terms of local reductions in the layers' elastic properties. 250



Figure 3: Cantilever layered Timoshenko beam model representing a masonry tower.

In a similar way to the uniform TB, harmonic excitations are assumed in such a way that u(z, t), $\theta(z, t)$, M(z, t)and V(z, t) can be written as:

$$u(z,t) = U(z) e^{i\omega t}, \quad \theta(z,t) = \Theta(z) e^{i\omega t},$$

$$M(z,t) = \mathcal{M}(z) e^{i\omega t}, \quad V(z,t) = \mathcal{V}(z) e^{i\omega t}.$$
(26)

²⁵³ Considering the equilibrium of a differential beam element (see Fig. 2), the state of every layer can be described
 ²⁵⁴ by the following matrix equation:

$$\frac{\partial \mathbf{f}(z)}{\partial z} = \mathbf{B} \, \mathbf{f}(z), \tag{27}$$

where $\mathbf{f}(z) = [U(z), \Theta(z), \mathcal{M}(z), \mathcal{V}(z)]^{\mathrm{T}}$ denotes the stress-displacement vector or state vector, and matrix **B** takes the form [33]:

$$\mathbf{B} = \begin{bmatrix} 0 & 1 & 0 & \frac{1}{\kappa GA(1-i\omega\mu)} \\ 0 & 0 & -\frac{1}{EI(1-i\omega\mu)} & 0 \\ 0 & \rho I \omega^2 & 0 & 1 \\ -\rho A \omega^2 & 0 & 0 & 0 \end{bmatrix}.$$
 (28)

According to the propagator matrix theory [57], it can be proved that if matrix **B** is a continuous function of *z*, as it is the case within each layer of the TB, the state vector $\mathbf{f}(z)$ at a given point z_o in a certain layer can be propagated throughout the layer as $\mathbf{f}(z) = \mathbf{P}(z, z_o) \mathbf{f}(z_o)$, where $\mathbf{P}(z, z_o)$ is the so-called propagator matrix from z_o . Solving the ordinary differential equation (27), $\mathbf{P}(z, z_o)$ can be written as:

$$\mathbf{P}(z, z_o) = e^{\mathbf{B}(z - z_o)}.$$
(29)

The propagator can be further decomposed using a Taylor series expansion to compute the exponential in Eq. (29). Nevertheless, it is more convenient to diagonalize the matrix **B** by a similarity transformation as **B** = **SAS**⁻¹, where **A** is the diagonal matrix of the eigenvalues of **B**, and **S** contains the corresponding eigenvectors by columns. It can be seen that the eigenvalues of **B** correspond to $\pm ik_1$ and $\pm ik_2$ [33], with the wave-numbers k_1 and k_2 given by Eqs. (17) and (18). In this way, Eq. (29) can be decomposed as [33]:

$$\mathbf{P}(z, z_o) = \mathbf{S} e^{\mathbf{\Lambda}(z-z_o)} \mathbf{S}^{-1},$$
(30)

where the exponential of a diagonal matrix is directly given by the exponential of the diagonal elements, that is $e^{A(z-z_o)} = \text{diag}\left(e^{ik_1(z-z_o)}, e^{-ik_1(z-z_o)}, e^{ik_2(z-z_o)}, e^{-ik_2(z-z_o)}\right)$. Given that $\mathbf{f}(z)$ must be continuous at the layer interfaces, the solution at an arbitrary interface z_k can be obtained as:

$$\mathbf{f}(z_k) = \prod_{i=0}^{k-1} \mathbf{P}(z_{i+1}, z_i) \, \mathbf{f}(z_o). \tag{31}$$

In view of the boundary conditions previously shown in Eqs. (13) and (14), the state vector is constrained at z = 0 and z = H as:

$$\mathbf{f}(0) = \begin{bmatrix} U_g \\ 0 \\ M_g \\ V_g \end{bmatrix}, \quad \mathbf{f}(H) = \begin{bmatrix} U_r \\ \Theta_r \\ 0 \\ 0 \end{bmatrix}.$$
(32)

Assuming that the base motion U_g is prescribed, the base bending moment M_g and shear V_g , as well as the top displacement U_r and rotation Θ_r , remain unknown. Their values are obtained by solving the determined linear system of four equations formed by the solution at the base and its propagation to the roof. Thereafter, the state vector of the system can be obtained at any height *z* by the propagator matrix approach in Eq. (31). Finally, once the bending moment $\mathcal{M}(z)$ is known, the bending-induced normal strain in the frequency domain $\Upsilon(z)$ can be obtained as:

$$\Upsilon(z) = \frac{\mathcal{M}(z)}{EI(1 - i\omega\mu)}.$$
(33)

277 3.3. Transfer Functions and Impulse Response Functions

Once the solution in the frequency domain is known, the system TFs can be readily computed as previously indicated in Section 2. If we assume the TB as a linear system with ground motion U_g (or ground acceleration, $-\omega^2 U_g$) as input, and transverse displacements U(z) and strains $\Upsilon(z)$ as outputs, the TFs can be obtained as $U(z)/U_g$ and $\Upsilon(z)/U_g$, respectively. More generally, the TFs can be defined between the motion of the structure at height z with respect to a reference level z_{ref} , that is to say, considering a virtual source at z_{ref} that does not necessarily coincide with the actual physical source. Let \hat{h}_u and \hat{h}_{ε} denote the TFs in terms of displacements and strains, respectively, as follows:

$$\hat{h}_u(z, z_{ref}, \omega) = \frac{\hat{u}(z, \omega)}{\hat{u}(z_{ref}, \omega)} = \frac{U(z, \omega)}{U(z_{ref}, \omega)},$$
(34)

285

$$\hat{h}_{\varepsilon}(z, z_{ref}, \omega) = \frac{\hat{\varepsilon}(z, \omega)}{\hat{u}(z_{ref}, \omega)} = \frac{\Upsilon(z, \omega)}{U(z_{ref}, \omega)}.$$
(35)

In order to avoid numerical instability due to division by null numbers, TFs are often regularized as:

$$\hat{h}_{u}(z, z_{ref}, \omega) \approx \frac{U(z, \omega) \overline{U(z_{ref}, \omega)}}{\left| U(z_{ref}, \omega) \right|^{2} + \eta},$$
(36)

287

$$\hat{h}_{\varepsilon}(z, z_{ref}, \omega) \approx \frac{\Upsilon(z, \omega) \overline{U(z_{ref}, \omega)}}{\left| U(z_{ref}, \omega) \right|^2 + \eta},$$
(37)

where the bar indicates complex conjugate, and η denotes a regularization parameter to avoid numerical instability.

In this work, we use $\eta = 0.1\overline{P}$ with \overline{P} being the average power of the reference input. The corresponding IRFs, $h_u(z, z_{ref}, t)$ and $h_{\varepsilon}(z, z_{ref}, t)$, are defined in the time domain and can be computed as the inverse Fourier transform

of the TFs as $h_u(z, z_{ref}, t) = \mathscr{F}^{-1}\{\hat{h}_u(z, z_{ref}, \omega)\}$ and $h_\varepsilon(z, z_{ref}, t) = \mathscr{F}^{-1}\{\hat{h}_\varepsilon(z, z_{ref}, \omega)\}$. Typically, IRFs can be only obtained for a finite frequency band $|\omega| < \omega_{max}$, that is:

$$h_u(z, z_{ref}, t) = \frac{1}{2\pi} \int_{-\omega_{max}}^{\omega_{max}} \hat{h}_u(z, z_{ref}, \omega) e^{-i\omega t} d\omega,$$
(38)

293

$$h_{\varepsilon}(z, z_{ref}, t) = \frac{1}{2\pi} \int_{-\omega_{max}}^{\omega_{max}} \hat{h}_{\varepsilon}(z, z_{ref}, \omega) e^{-i\omega t} d\omega.$$
(39)

294 **4. Results and discussion**

This section presents numerical results and discussion on the application of acceleration- and strain-based wave propagation analysis for damage identification in masonry towers under seismic actions. Specifically, the analyses are divided into: (i) parametric analyses of travelling waves on the basis of the developed TB model in Section 4.1, and (ii) validation case study of a 41.6 m high masonry tower in Section 4.2. In the latter, a nonlinear 3D FEM of the tower has been developed and subjected to synthetic base acceleration series. The IRFs are computed on the basis of recorded acceleration and strain signals at different heights of the tower, and the results are compared with the proposed layered TB model for damage identification.

302 4.1. Parametric analyses

In this set of analyses, numerical results are first presented to illustrate the structure of the dispersion relation 303 in Eq. (15). Without loss of generality, masonry towers with square cross-section and shear correction factor 304 κ =0.43 (ν = 0.25) [58] are investigated. As previously reported by Ebrahimian and Todorovska [32], two different 305 branches of dispersion curves can be distinguished corresponding to wavenumbers k_1 and k_2 in Eqs. (17) and 306 (18). These represent two different wave propagation modes, and each mode can be up-going (plus sign outside) 307 and down-going (minus sign outside). A closer inspection of Eqs. (17) and (18) reveals that k_1 is real-valued 308 for all Ω , while k_2 only becomes real when $\Omega > \Omega_{cr} = \sqrt{\kappa}$ [32], with Ω_{cr} being the cut-off frequency for the 309 second propagation mode. When $\Omega < \Omega_{cr}$, k_2 is complex-valued and, thus, the second propagation mode defines 310 exponentially attenuated non-propagating waves, also referred to as near field or evanescent waves [59]. In order 311 to elucidate the physical mechanisms underlying both propagation modes, Fig. 4 depicts the ratio of bending 312 moment to shear force $|\mathcal{M}(z)|\mathcal{V}(z)|$ of the undamped system $\mu = 0.0$ for each mode at height z = H, and different 313 moduli ratios, namely R = 0.01, 0.50 and 1.00. Firstly, it is observed that for larger moduli ratios, R = G/E, 314 the contribution of bending moment to the propagation modes becomes more predominant. It is also observed 315 that the first propagation mode is dominated by bending moment at low frequencies below the cut-off frequency 316 $\Omega < \Omega_{cr}$, whereas the contribution of shear increases at higher frequencies $\Omega > \Omega_{cr}$. On the other hand, the second 317 propagation mode is always dominated by bending, except around the cut-off frequency Ω_{cr} where the bending 318 moment approaches zero and shear dominates. This limit corresponds to a thickness-shear mode of the beam, 319 which is characterized by pure distortion of the cross section with zero deflection. 320



Figure 4: Bending-shear ratio, $|\mathcal{M}(z)|\mathcal{V}(z)|$, versus non-dimensional frequency $\Omega = \omega r_g/c_S$ for harmonic waves in a Timoshenko beam.

The behaviour of the propagation modes in terms of strains is further investigated in Fig. 5. For clarity purposes, $\Upsilon(z)$ in Eq. (25) can be expanded as:

$$\Upsilon(s) = D_1 \cos k_1 s + D_2 \sin k_1 s + D_3 \cos k_2 s + D_4 \sin k_2 s, \tag{40}$$

with s = H - z, and D_i the coefficients of expansion given as:

$$D_{1} = E_{1} \cos k_{1}H + E_{2} \sin k_{1}H, \quad D_{2} = E_{1} \sin k_{1}H - E_{2} \cos k_{1}H,$$

$$D_{3} = E_{3} \cos k_{2}H + E_{4} \sin k_{2}H, \quad D_{4} = E_{3} \sin k_{2}H - E_{4} \cos k_{2}H,$$
(41)

and the terms E_i as:

$$E_{1} = -ik_{1}B_{1} + ik_{1}B_{2}, \quad E_{2} = k_{1}B_{1} + k_{1}B_{2},$$

$$E_{3} = -ik_{2}B_{3} + ik_{2}B_{4}, \quad E_{4} = k_{2}B_{3} + k_{2}B_{4}.$$
(42)

Figures 5 (a, c, e) show the coefficients of expansion D_i , $i = 1, \dots, 4$, normalized by C_1 for a fixed value of 325 R = 0.5 and different ratios $H/r_g = 4, 8$, and 16. In addition, the module of the TFs in terms of acceleration 326 $|TF_a|$ is also depicted in Figs. 5 (b, d, f) to illustrate the number of natural modes of vibration contained in the 327 waveforms. For clarity purposes, the expansion coefficients are computed for undamped conditions, while the TFs 328 are calculated considering different dimensionless damping constants, namely M = 0.0, 0.001 and 0.002. Firstly, 329 note that D_1 equals D_3 because of the boundary condition M(H) = 0.0. It is seen that, for larger H/r_g ratios, 330 i.e. slenderer towers, the natural modes of vibration have lower frequencies and are more affected by dispersion. 331 In addition, a higher number of natural modes of vibration fall in the frequency interval $\Omega < \Omega_{cr}$. The effect of 332 dispersion can be inferred in the analysis of the coefficients of expansion D_i in Figs. 5 (a, c, e). It is observed in 333 these figures that, below the critical frequency Ω_{cr} , the variation of the coefficients of expansion D_i with frequency 334 is smooth. At the critical frequency Ω_{cr} , the wavenumber k_2 of the second wave propagation mode in Eq. (18) tends 335 to zero and, as a result, so does the coefficient D_4 . Beyond Ω_{cr} , the coefficients of expansion D_i change very fast 336 with frequency, what suggests the presence of complex interference patterns of the two wave propagation modes. 337 This effect is accentuated for slenderer towers, i.e. larger H/r_g ratios, where faster variations of D_i with frequency 338 are observed for frequencies above Ω_{cr} . Similar conclusions can be extracted in terms of lateral displacements as 339 reported by Ebrahimian and Todorovska [32]. 340



Figure 5: Magnitudes of the coefficients of expansion D_i , i = 1, ..., 4, normalized by the coefficient C_1 , versus dimensionless frequency $\Omega = w r_g/c_s$ (a,c,e), and the acceleration TF amplitudes $|TF_a|$ (roof with respect to ground) (b,d,f).

Figures 6 and 7 present a parametric study of the IRFs versus dimensionless time $\bar{t} = t c_S / r_g$. Firstly, Fig. 6 341 investigates different moduli ratios R = 0.01, 0.5, and 1.0, for a virtual source at the roof $z_{ref} = 0$ (a, d, g) and at 342 the base $z_{ref} = H$ (b, e, h), assuming a constant ratio $H/r_g = 8$ and a dimensionless damping constant M = 0.05. 343 Secondly, Fig. 7 analyzes varying ratios $H/r_g = 6$, 11, and 16, considering a constant moduli ratio R = 0.5 and a 344 dimensionless damping constant M = 0.05. In addition, the TFs in terms of accelerations $|TF_a|$ and strains $|TF_s|$ 345 are also shown (c,f,i) versus dimensionless frequency Ω . To do so, the TFs in terms of accelerations are computed 346 by the ratio of accelerations at the roof level to the ground acceleration, while the TFs in terms of strains are 347 obtained by the ratio of strains at the ground level to the ground acceleration. In these figures, the acceleration 348 and the strain waves are shown with dashed and solid lines, respectively. Also, some ray paths are marked in 349 red to indicate travelling pulses. In order to avoid complex waveforms due to the interaction of the second wave 350 propagation mode, the pulses have been low-pass filtered at $\Omega = 0.66 \approx \Omega_{cr}$. In can be noted in Figs. 6 (a, d, 351 g) and Figs. 7 (a, d, g) that the waveforms considering a virtual source at the roof level show two propagating 352 pulses, typically termed causal and acausal pulses. As reported by Snieder and Safak [23], the consideration of 353 such virtual sources imposes a condition of zero roof motion at all times except during the application of the 354 source and, therefore, reflections from the base are suppressed. Conversely, wave propagation for virtual sources 355 at the base (b,e,h) only shows causal pulses with multiple reflections at the base and the roof level. In this case, 356 a clear identification of travelling pulses is often an intricate task, whereby the use of virtual sources at the roof 357 level is usually more convenient for system identification purposes. In the case of acceleration waves and virtual 358 sources at the roof level, it is noted that dominant downward propagating causal and acausal pulses can be clearly 359 identified in all the cases. At the roof level, the source pulse computed by Eq. (38) for $z = z_{ref} = H$ is always a sinc 360 function $\sin \omega_{max} t/\pi t$. Hence, the effect of dispersion is evidenced by the deformation of the source pulse when 361 propagating throughout the tower. In particular, it is observed that larger ratios R and H/r_g lead to more dispersive 362 systems. On the other hand, a noticeably different behaviour can be observed in the strain waves. The strain-free 363 condition of the cantilever TB model, $\Upsilon(H) = 0$, precludes the development of any pulse at the roof level. As 364 a result, the appearance of causal and acausal pulses requires longer distances, specially for smaller ratios R and 365

- $_{366}$ H/r_g . This fact hinders the applicability of strain-based wave propagation analysis for the identification of regions
- of the structure close to the roof level, as can be observed in the case of R = 0.01 where no pulse can be identified.



Figure 6: IRFs at different levels in the beam for virtual source at the roof (a,d,g), and at the base (b,e,h), versus dimensionless time $\bar{t} = tc_S/r_g$, and TFs in terms of accelerations $|TF_a|$ and strains $|TF_{\varepsilon}|$ (c,f,i) versus dimensionless frequency Ω , considering different moduli ratios R = 0.01, 0.5, 1.0, and constant parameters $H/r_g = 8, M = 0.05$.



Figure 7: IRFs at different levels in the beam for virtual source at the roof (a,d,g), and at the base (b,e,h), versus dimensionless time $\bar{t} = tc_S/r_g$, and TFs in terms of accelerations $|TF_a|$ and strains $|TF_{\varepsilon}|$ (c,f,i) versus dimensionless frequency Ω , considering different ratios $H/r_g = 6$, 11, 16, and constant parameters R=0.5, M = 0.05.

Finally, some results are presented regarding the influence of dispersion on the phase and group velocities. According to their definition, namely $c^p = \omega/k$ and $c^g = \partial \omega/\partial k$, along with the dispersion relation in Eq. (15), analytical expressions of the phase and group velocities can be obtained as follows [32]:

$$c_{1,2}^{ph} = \frac{c_S \,\Omega}{k_{1,2} r_g},\tag{43}$$

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$$c_{1,2}^{gr} = \frac{c_S}{r_g \partial k_{1,2} / \partial \Omega} = 2c_S r_g k_{1,2} \left\{ \Omega \left[\left(\frac{1}{\kappa} + R \right) \pm \frac{\left(\frac{1}{\kappa} - R \right)^2 + \frac{2R}{\Omega^2}}{\sqrt{\left(\frac{1}{\kappa} - R \right)^2 + \frac{4R}{\Omega^2}}} \right] \right\}^{-1}.$$
 (44)

An inspection of Eqs. (43) and (44) reveals that when $R \le 1/\kappa$, as it is typically the case, $\lim_{\Omega \to \infty} c_1^{ph} = \lim_{\Omega \to \infty} c_1^{gr} = 1$ 372 $c_S \sqrt{\kappa}$ and $\lim_{\Omega \to \infty} c_2^{ph} = \lim_{\Omega \to \infty} c_2^{gr} = c_S / \sqrt{R}$. Figure 8 shows the phase and group velocities of the first propagating waves versus dimensionless frequency Ω considering varying moduli ratios *R*. In this figure, two different shear 373 374 correction factors are selected, namely $\kappa = 0.43$ (a) and $\kappa = 0.85$ (b), which correspond to TBs with thin-375 walled hollow square and full rectangular cross-sections, respectively, according to Cowper's formulae ($\nu = 0.25$) 376 [58]. In these plots, the critical cut-off frequency Ω_{cr} is also marked. Let us recall that, in order to avoid the 377 cumbersome interference of the two wave propagating modes, the frequency band of interest limits to $\Omega < \Omega_{cr}$ 378 where only evanescent waves are given by the second propagation mode. Given that $\omega_{cr} = \sqrt{\kappa c_s} / r_g$, the frequency 379 band of interest is wider when c_s is larger (i.e. larger shear modulus G or smaller mass density ρ), r_g is smaller 380 (i.e. larger cross-section area A or smaller inertia I), and κ is larger. The shear correction factor κ amends the 381

effect of uniform shear stresses in the cross-section, and smaller factors lead to more flexible structures with lower natural frequencies. It is observed in Fig. 8 that larger ratios *R* lead to more dispersive structures and, as a result, the deviation of the phase/group velocities from their asymptotic dimensionless value $\sqrt{\kappa}$ increases. Conversely, small values of *R* yield phase/group velocities that stabilize for low frequencies and, therefore, the resulting system is less dispersive. Finally, it is observed that increasing shear correction factors κ lead to larger cut-off frequencies Ω_{cr} and, as a consequence, more dispersive structures.



Figure 8: Dimensionless phase velocity, c_1^{ph}/c_s , and group velocity, c_1^{gr}/c_s , of mode 1 propagating waves versus dimensionless frequency, $\Omega = w r_g/c_s$, for varying moduli ratios R = 0.01, 0.1, 0.5 and 1, and shear correction factors $\kappa = 0.43$ (a) and $\kappa = 0.85$ (b).

388 4.2. Validation case study

In the remainder of this paper, a case study of a masonry tower is presented in order to validate and further 389 investigate the potentials of the proposed methodology. This consists of the 41.6 m high civic tower located in the 390 historical centre of Perugia in Italy, named Torre degli Sciri. The tower can be ideally divided into two structural 391 portions with geometrical dimensions shown in Fig. 9 (a). The lower part is characterized by a hollow rectangular 392 cross-section with wall thicknesses of 1.68 m and 2.1 m and rises up to 8.8 m. There are some small openings 393 and a stone masonry vaulted slab that stands above the rooms of an old chapel. On the other hand, the upper 394 part rises up to 41 m and has slender 1.68 m thick continuous walls, with four 1.5 m wide brick masonry vaulted 395 slabs at different heights. Moreover, a brick masonry ceiling vault completes the tower on the top and a 0.5 m 396 thick parapet extends up to a height of 41.6 m. The masonry is homogeneous and regular, and it is made of 397 squared white limestone blocks. Although the tower is incorporated into a building aggregate, the isolated tower 398 is considered as a case study in this paper for a clear comparison against the Timoshenko beam model. 399



Figure 9: Case study of the Sciri Tower: (a) geometrical dimensions, (b) FEM of the tower and (c) sensors arrangement (units in meters).

In order to reproduce different earthquake-induced damage scenarios and validate the proposed damage iden-400 tification approach, a non-linear 3D numerical model of the Sciri Tower under ground accelerations has been built 401 in the framework of the FEM by using ABAQUS 6.10 platform [60]. The model, shown in Fig. 9 (b), has been 402 constructed on the basis of information gained from available structural drawings. The floors and walls of the 403 tower have been included in the model and the whole structure has been assumed fixed to the ground, considering 404 the soil-structure interaction as negligible. A free meshing of solid C3D4 tetrahedral elements with mean elements' 405 dimension of about 40 cm has been adopted. The elastic properties of the FEM materials are assumed isotropic, 406 with elastic parameters computed according to the Italian technical standard [61] for square stone masonry, in-407 cluding Young's modulus E = 5.77 GPa, Poisson's ratio $\nu = 0.25$, and mass density $\rho = 2.2$ t/m³. The non-linear 408 behaviour of masonry is modelled with the classic Concrete Damage Plasticity (CDP) constitutive model [60]. 409 This approach, proposed by Lubliner et al. [62] and then modified by Lee and Fenves [63], is well-suited for 410 the modelling of brittle masonry under cyclic loading considering cracking in tension and crushing in compres-411 sion [64, 65]. Given the lack of characterization tests of the masonry of the tower, the non-linear mechanical 412 properties assigned to the FEM have been estimated from the literature as shown in Table 1. Preliminary re-413 sults showed no compression damages in the structure, whereby, for simplicity, the material is assumed elastic in 414 compression and brittle in tension. Regarding the seismic loading, Eurocode 8 spectrum-compatible synthetic ac-415 celerograms have been generated and applied in the x-direction of the tower (see Fig. 9 (b)). Three different Peak 416 Ground Accelerations (PGAs) have been considered in order to analyse distinct damage severities, including 0.1g, 417 0.15g and 0.2g. In particular, 70 s long time series have been obtained with a time sampling frequency $F_s = 100$ 418 Hz. In order to account for the non-stationarity of the seismic events, the steady state ground accelerations have 419 been modulated by a compound intensity envelope [66] with rise time of 10 s and decay time of 40 s, resulting in a 420 strong-motion duration of 30 s. The time history analysis of the dynamic response of the tower has been conducted 421 by the Hilbert-Hughes-Taylor implicit direct time integration scheme, accounting for material non-linearities with 422 the full Newton-Raphson method. Structural damping has been considered by the classical Rayleigh formulation, 423 with 5% of damping ratios on the first two modes. 424

Horizontal accelerations and vertical strains are monitored along the height of the tower FEM. In addition, two different Timoshenko beam models have been studied, namely a uniform TB and a three-layered TB, labelled with U-TB and 3L-TB, respectively. The three-layered Timoshenko beam (Fig. (9) (c)) considers three sections of the tower rising from 0-8.8 m (L1), 8.8-25.9 m (L2) and 25.9-41.6 m (L3).

Table 1: Non-linear cor	npressive and	tensile behaviour	of the CDF	P model of masonry
				2

	Elasto-plast	ic behaviour	Tensile behaviour			
Ε	5.77 GPa $K [s]^a$		0.67	Stress [kN/m ²]	Damage parameter d_t	
ν	0.25	Viscosity parameter ^b	0.00	160	0.00	
ho	2.2 t/m ³			120	0.55	
Dilation angle	21°			84	0.80	
Eccentricity	0.10			16	0.90	

^a K is the ratio of the second stress invariant on the tensile meridian.

^b The viscosity parameter is used for the viscoplastic regularization of the constitutive equations.

Analyses and discussion have been divided into system identification, damage identification by inverse cal-429 ibration of the TB model, and data-driven damage identification approach in Sections 4.2.1, 4.2.2 and 4.2.3, 430 respectively. Firstly, the presented results are intended to show the suitability of the developed TB model to repro-431 duce the dynamic response of the undamaged tower in the frequency domain, as well as the travelling waves and 432 their relationship with bending and shear stiffness. Afterwards, the correlation between the earthquake-induced 433 damages and the speed of the travelling waves is investigated. To do so, two different damage identification ap-434 proaches are presented, including a model-based and a data-driven approach. The first one relates the inverse 435 calibration of the TB model, while the second one regards the peak-picking analysis of wave travel times directly 436 obtained from seismic records. 437

438 4.2.1. System identification

The response time series are divided in three time windows - before, during, and after the strong-motion, that 439 is $0 \le t < 12 \le 12 \le t < 40 \le$ and $40 \le t < 70 \le$. This first set of analyses focuses on the first time window, 440 which serves as a reference baseline. The baseline is characterized by the absence of damages, that is to say, the 441 structure fully remains in its elastic regime. In particular, a PGA of 0.1g is selected, and the time histories of the 442 recorded accelerations and strains used in this study are shown in Fig. 10 (a). In order to identify the structural 443 system, the three-layered TB is updated by minimizing the root-mean-squared error between the analytical IRFs 444 at the layers' interfaces and those obtained by the post-processing of the recorded time series. The resulting non-445 linear minimization problem is highly ill-conditioned, thereby a Particle Swarm optimization algorithm is used to 446 fit the model parameters. To reduce the number of unknown parameters and have a more robust fit, only two of the 447 layer parameters are fitted, namely the shear wave velocity c_s and moduli ratio R, while the rest are estimated from 448 the geometry or assumed. A shear correction factor of $\kappa = 0.46$ and a small value of the Kelvin-Voigt damping 449 constant $\mu = 2.5$ E-3 are assumed constant all along the tower so that the first few modes are visible in the TFs. 450 The cross-section A, inertia I and mass density ρ are estimated from the geometry as furnished in Table 2, and the 451 Poisson's ratio v = 0.25 is assumed constant. In the case of the uniform Timoshenko beam (U-TB), the radius of 452 gyration is $r_g = 2.4$ m, the critical frequency for the model $f_{cr} = 45.95$ Hz, and the shear-wave velocity in the 453 material $c_s = 1018.97$ m/s. 454

Table 2: Values of the parameters of the beam models that are estimated from the geometry or assumed. Labels U-TB and 3L-TB stand for uniform Timoshenko beam and three-layered Timoshenko beam, respectively.

					UTB		3L-TB			
Layer	<i>z</i> [m]	<i>h</i> [m]	μ[-]	ρ [kg/m ³]	<i>A</i> [m ²]	<i>I</i> [m ⁴]	ρ [kg/m ³]	<i>A</i> [m ²]	<i>I</i> [m ⁴]	
L1	0.0-8.8	8.84	2.50E-03	2222.9	38.40	220.07	2245.9	41.69	224.25	
L2	8.8-25.9	17.09	2.50E-03	2222.9	38.40	220.07	2155.6	38.40	220.07	
L3	25.9-41.6	15.67	2.50E-03	2222.9	38.40	220.07	2174.9	38.40	220.07	

The transfer functions computed by the 3D FEM and the Timoshenko beam models in terms of accelerations 455 $(|TF_a|)$ and strains $(|TF_{\varepsilon}|)$ are compared in Figs. 10 (b) and (c), respectively. In the acceleration TFs, only three 456 clear peaks corresponding to the first three natural modes can be observed. Similarly, the strain TFs exhibit two 457 clear peaks at the first resonant frequencies, while the third peak is considerably attenuated and the TFs are notice-458 ably noisy at higher frequencies due to poor frequency sampling. In this case, the U-TB suffices to estimate the 459 modes activated by the earthquake and, as a result, little improvement of the fitted 3L-TB is noted. Then, Figure 11 460 shows the IRFs computed by the 3D FEM and the TB models. Specifically, comparisons of the IRFs in terms of 461 accelerations and strains with a virtual impulse at the roof level are shown considering different frequency bands, 462 namely 0.8-10 Hz and 0.8-20 Hz. In accordance with the TFs previously shown in Fig. 10 (b), these frequency 463

bands contain up to the second and the third vibration modes, respectively. It should be noted that the signals 464 are band-filtered with a low cut-off frequency of 0.8 Hz to eliminate the low-frequency contributions stemming 465 from damage-induced load path changes, specially for strain signals in the second and third time windows. The 466 unknown layer parameters of the 3L-TB are estimated by fitting the IRFs in the band of frequencies 0.8-20 Hz, 467 and the resulting fitted parameters are shown in Table 3. Let us remark that the moduli ratio R = G/E corresponds 468 to the structural layer as a whole and not simply to the material. While qualitative good agreement can be observed 469 in Fig. 11 between the IRFs computed by the FEM and the U-TB, closer fittings are observed for the 3L-TB and, 470 thus, the latter model provides a better representation of the pulse propagation throughout the tower. More specif-471 ically, mean coefficients of determination R^2 (averaged over the IRFs at the layers' interfaces) of 0.84, 0.78, 0.64 472 and 0.74 are obtained for the U-TB in Figs. 11 (a), (b), (c) and (d), respectively, while the respective values for 473 the 3L-TB are 0.78, 0.80, 0.77 and 0.79. Note that the travelling waves in Figs. 11 (c,d) move much faster than 474 those in Figs. 11 (a,b). This fact is indicative of a meaningful contribution of dispersion, which yields faster waves 475 at higher frequencies. It is also observed that, while acceleration waves show the first causal and acausal pulses 476 at z = 29.3 m and 32.8 m for frequency bands of 0.8-10 Hz and 0.8-20 Hz, respectively, identifiable travelling 477 strain waves cannot be observed until z = 5.4 m and 19.0 m for frequency bands of 0.8-10 Hz and 0.8-20 Hz, 478 respectively. This behaviour is due to the strain distribution in a building clamped at the base, which is zero at the 479 roof level and, as a consequence, travelling waves require larger distances to develop. Therefore, it is concluded 480 that monitoring approaches based on acceleration records are efficient for the identification of travelling waves at 481 low and moderate heights, while those based on strain measurements are limited to low heights. Finally, let us 482 point out the presence of considerable differences between the strain IRFs at z = 8.8 m. In this case, the sensor 483 is located close to an opening (see Fig. 9 (c)) and the strain series is highly conditioned by local stiffness effects. 484 Notwithstanding the circumstance that beam models fail to represent such local effects on strain and, as a result, 485 local wave attenuation cannot be properly reproduced, it is observed in Figs. 11 (b,d) that the propagation of the 486 strain waves is very similar. 487



Figure 10: Horizontal accelerations and vertical strains monitored at different heights of the FEM of the Sciri tower ($F_s = 100$ Hz) under ground accelerations of PGA=0.1g (a), and comparison of the TFs computed by the FEM and the beam models in terms of accelerations $|TF_a|$ (b), and vertical strains $|TF_{\varepsilon}|$ (c). Labels U-TB and 3L-TB stand for uniform Timoshenko beam and three-layered Timoshenko beam, respectively.



Figure 11: Comparison of the IRFs in terms of accelerations (a,c) and strains (b,d) computed by the three-dimensional FEM, uniform Timoshenko beam (U-TB) and fitted three-layered Timoshenko beam (3L-TB), considering frequency bands of 0.8-10 Hz (a,b) and 0.8-20 Hz (c,d), and a virtual impulse source at the roof level $z_{ref} = 41.6$ m.

Table 3: Initial and fitted shear-wave velocities c_s and moduli ratios R on the frequency band 0.8-20 Hz.

	Initial			Fitted			
Layer	<i>cs</i> [m/s]	R [-]		$c_S \text{ [m/s]}$	R [-]		
L1	1.030E+03	0.40	1	1.046E+03	0.47		
L2	1.035E+03	0.40	ç	9.662E+02	0.45		
L3	1.014E+03	0.40	7	7.946E+02	0.27		

Finally, Fig. 12 shows the distributions of bending stiffness *EI* and shear stiffness *GA* for the initial and fitted 3L-TB models along the height of the tower. Overall, it is observed that the fitted 3L-TB yields a stiffness distribution that decreases with height. It is interesting to note that the bending stiffness of the second layer L2 is slightly lower than that of the third layer L3. The reason for this is ascribed to the presence of a higher concentration of openings in the second layer L2 as can be observed in Fig. 9 (c). All things considered, it is concluded that the developed Timoshenko beam model is suitable for the structural identification of masonry towers.



Figure 12: Bending stiffness EI (a) and shear stiffness GA (b) for the uniform Timoshenko beam (U-TB) and fitted 3-layered Timoshenko beam (3L-TB) on the frequency band 0.8-20 Hz.

495 4.2.2. Damage identification: Inverse TB calibration

In this second set of results, wave propagation analyses are conducted for increasing damage severities, includ-496 ing PGA values of 0.1g, 0.15g and 0.2g. The computed damage patterns after the considered seismic events are 497 depicted in Fig. 13 in terms of the tensile damage factor d_t . This factor characterizes the degradation of the elastic 498 tensile stiffness of the masonry and takes values between 0 (undamaged) and 1 (fully damaged) [60]. It is observed 499 that damages concentrate in the bottom part of the tower. In the case of low PGA of 0.1g, damages mainly localize 500 around z = 8.8 m where the wall thickness diminishes and there is a concentration of openings, as well as in the 501 right façade at x = 7.27 m. As the PGA increases, damages first propagate in the left/right façades at x = 0.0, 7.27502 m and, afterwards, in the front/back façades at y = 0.0, 7.52 m. These damage patterns are primarily determined 503 by the first bending mode, including a set of horizontal cracks located in the left/right façades at x = 0.0, 7.27 m, 504 as well as shear X-cracks starting at the corners of the openings located in the front/back façades at y = 0.0, 7.52505 506 m.



Figure 13: Computed crack patterns in the 3D FEM of the masonry tower under seismic ground accelerations in the x-direction considering increasing PGAs, namely (a) 0.1g, (b) 0.15g and (c) 0.2g. The parameter d_t stands for the tensile damage and represents local earthquake-induced tensile stiffness degradation.

Figure 14 shows the seismic ground accelerations (a1, b1, c1), the IRFs in terms of accelerations (a2, b2, c2) 507 and strains (a3, b3, c3) at the layers' interfaces of the 3L-TB (z=0.0 m, 8.8 m, 25.9 m, and 41.6 m) considering a 508 virtual impulse at the roof level, and the TFs (a4, b4, c4) for PGAs of 0.1g (a), 0.15g (b) and 0.2g (c). The plots in 509 Fig. 14 show that the propagation velocities of the pulses for the three time windows are different, indicating longer 510 travel times (i.e. reduced stiffness) within the second and third time windows as compared to the first one (see 511 inserts in Figs. 14 (c2) and (c3)). It is also observed that the resonant peaks in the TFs experience slight shifts, 512 what also indicates earthquake-induced stiffness losses. For instance, the fundamental frequency (first bending 513 mode shown in Fig. 10 (b)) shifts from 1.20 Hz to 1.18 Hz, 1.16 Hz and 1.15 Hz for PGAs of 0.1g, 0.15g and 0.2g, 514 respectively. In order to identify the earthquake-induced damages, the three-layered TB model is updated with the 515 results obtained by the FEM in a similar way to Section 4.2.1. As a result, Table 4 furnishes the fitted shear wave 516 velocities c_s , moduli ratios R, and Normalized Root-Mean-Squared Error (NRMSE). The error is normalized by 517 the amplitude of the numerical IRFs, and the global NRMSE corresponds to the mean value of the errors at the 518 layers' interfaces. It is noted in Table 4 that the NRMSE increases with the PGA, although it remains below 5% 519 except for the case of strain waves with PGA=0.2g. The reason for such an increase is ascribed to the increasing 520 damage levels, which propagate non-symmetrically and the hypotheses of the TB model become less realistic. In 521 addition, it is observed that the NRMSE is larger for strain IRFs, what is due to local effects not included in the 522 TB model. 523

On the basis of the previous results, Figs. 15 (a) and (b) show the variations of the bending and shear stiffness 524 distributions along the height of the masonry tower, respectively. For clarity purposes, Fig. 15 (c) presents the 525 damage patterns computed by the 3D FEM in terms of damage factors d_t obtained on the front (y = 0.0 m) and 526 right façades (x = 7.27 m) for every PGA value. Overall, it is observed that higher PGA values yield larger 527 stiffness losses and, therefore, these results demonstrate the capability of this approach to quantify damages. It 528 is also noted that the largest stiffness reductions are found in the bottom part of the tower. Specifically, bending 529 stiffness reductions in the first layer L1 are equal to 5.97%, 19.56% and 38.66% for PGAs of 0.1g, 0.15g and 0.2g, 530 respectively, while approximately constant reductions of 8.90% are found in the second layer L2. Finally, only 531

small spurious reductions below 1% are obtained in the third layer L3 due to some ill conditioning of the model

⁵³³ updating approach. In view of Fig. 15 (c), these variations are in good agreement with the computed earthquake-

induced damages and, therefore, it is concluded that the proposed methodology is suitable for damage detection,
 localization and quantification.



Figure 14: Ground accelerations, IRFs in terms of accelerations (IRF_a) and strains (IRF_c) (0.8-20 Hz), and TFs computed by the 3D FEM of the masonry tower considering different values of PGA, including 0.1g (a), 0.15g (b) and 0.2g (c).

Table 4: Fitted shear-wave velocities c_s and moduli ratios R on the frequency band 0.8-20 Hz, and Normalized Root-Mean-Squared Error (NRMSE). The term BL stands for the baseline or the undamaged condition of the structure.

$c_S [m/s]$					R [-]				NRMSE [%]		
PGA	L1	L2	L3		L1	L2	L3		Accel.	Str.	
BL	1.094E+03	9.604E+02	8.355E+02		0.468	0.363	0.338		2.44	3.82	
0.1g	1.090E+03	9.596E+02	8.347E+02		0.473	0.364	0.339		2.74	3.82	
0.15g	1.082E+03	9.594E+02	8.333E+02		0.475	0.364	0.339		2.97	4.48	
0.20	1.047E+03	9.473E+02	8.333E+02		0.475	0.366	0.339		3.10	6.18	



Figure 15: Variation of bending stiffness (*EI*) (a) and shear stiffness (*GA*) (b) with respect to their undamaged distributions (*EI*^o and *GA*^o) for the fitted three-layered Timoshenko beams (3L-TB) on the frequency band 0.8-20 Hz. (c) Front and lateral views of the computed crack patterns in the 3D FEM of the masonry tower.

Finally, Fig. 16 shows the phase and group velocities (Eqs. (43) and (44)) of the fitted 3L-TB model for 536 PGA=0.2g. Compared to the shear and bending stiffness distributions, dispersion curves offer a more complete 537 representation of the system since information is given in the whole studied frequency band. It is noted that the 538 phase and group velocities change substantially with frequency, what evidences a determinant role of dispersion 539 as previously observed in Fig. 11. This behaviour also persists for frequencies above the critical frequency (recall 540 that $f_{cr} = 45.95$ Hz for the uniform TB). Therefore, the theoretical asymptotes of the phase and group velocities, 541 that are $c_s \sqrt{\kappa}$ and c_s / \sqrt{R} , respectively, are not reached in the considered frequency band. Dispersion hinders the 542 identification of pulses since there is not a predominant wave, but the interference of waves with different phase 543 velocities that form a complex wavefront. In this light, and considering that the studied tower has geometrical 544 dimensions and material properties that are representative of most isolated masonry towers, the proposed Timo-545 shenko beam model offers a valuable tool for damage identification through the fitting of IRFS, accounting for the 546 scattering and attenuation of complete waveforms in a certain frequency band. 547



Figure 16: Phase c^{ph} and group c^{gr} velocities of propagating waves in the fitted 3L-TB model for PGA=0.2g and a frequency range of 0.8-20 Hz versus frequency (0 s < t < 12 s).

548 4.2.3. Damage identification: Data-driven approach based on the analysis of wave delays

One of the most remarkable features of seismic wave interferometry relates the possibility of conducting the 549 damage identification in a completely data-driven way. To do so, wave travel times τ_i and velocities $v_i = l_i / \tau_i$ can 550 be directly obtained by peak-picking the arrival times of the identified pulses t_i at the transducers' positions as 551 previously reported in Section 2. If this process is repeated for each time window, damages can be inferred from 552 changes relative to the first time window or baseline (BL). In this light, Table 5 reports the computed values of 553 τ_i and v_i by the peak-picking analysis of the IRFs obtained by the FEM and the 3L-TB model on the frequency 554 band of 0.8-20 Hz and the time window 40 s < t < 70 s. In addition, for clarity purposes, Fig. 17 summarizes 555 the obtained results in a graphical way. In this figure, the variations in wave velocities are presented in relative 556 terms as $100(1 - v_i/v_i^o)$, with v_i^o being the wave velocity obtained from the BL. It is noted that the maximum 557 decreases in the wave velocities are localized in the bottom part of the tower. In particular, the velocities of the 558 acceleration waves computed by the FEM experience decreases of 1.56% and 6.56% for PGAs of 0.15g and 0.2g, 559 respectively (see Fig. 17 (a)). In the first case of PGA=0.1g, this methodology fails to localize the damage, even 560 reporting a very small increase of 0.06% in the wave velocity. This can be ascribed to dispersion effects, as well 561 as to the selected sampling frequency of $F_s = 100$ Hz, which may be insufficient to detect very small variations 562 caused by mild damage levels. In the second layer L2, reductions of 0.45%, 0.44% and 0.55% are found for PGAs 563 of 0.1g, 0.15g and 0.2g, respectively, while only small variations below 0.2% are found in the third layer L3. 564 Moreover, it is possible to compute the global velocity of the waves, that is the velocity of the waves crossing the 565 whole tower (see Fig. 17 (c)). In this case, wave velocities of 462.74 m/s, 461.78 m/s, 460.61 m/s and 456.60 m/s 566 are obtained for the undamaged state (BL), and damaged states under seismic events with PGAs of 0.1g, 0.15g, 567 and 0.2g, respectively. Therefore, it is noted that, although some limitations may arise to precisely localize the 568 damage for low damage severities (e.g. PGA=0.1g), this approach always remains suitable for damage detection 569 and quantification. 570

In the case of strain waves, due to the free-strain condition at the roof level, identifiable travelling pulses 571 are not observed in the third layer L3 as previously shown in Fig. 14. Therefore, the computed wave velocity 572 in the second layer L2 in Table 5 corresponds to the velocity of the waves crossing the layers L2 and L3. In 573 this case, considerably high wave velocities are computed which, along with the limitations derived from the 574 free-strain condition, may be due to poor time sampling. Nonetheless, damages in the first layer L1 are well 575 detected with similar variations in the wave velocities to those reported for acceleration waves (see Fig. 17 (b)). 576 Furthermore, the analysis of the global wave velocities yields values of 585.38 m/s, 585.25 m/s, 581.18 m/s and 577 577.12 m/s for the undamaged state (BL), and damaged states with PGAs of 0.1g, 0.15g, and 0.2g, respectively. 578 It is thus concluded that, although more limited for damage localization, strain wave propagation approaches 579 remain suitable for damage detection and quantification in masonry towers. Additionally, in order to evaluate the 580 soundness of the proposed three-layered TB model, Table 5 also shows the Relative Error (RE) in the determination 581 of the wave velocities. It is observed that the RE remains below 9% in all the cases and, therefore, it is concluded 582 that the proposed approach is apt for characterizing the wave propagation in masonry towers. 583

Table 5: Measurements of wave travel times τ_i and average wave velocities v_i between layers' interfaces for the 3D FEM and the fitted three-layered TB (3L-TB) on the frequency band 0.8-20 Hz (40 s < t < 70 s, $F_s = 100$ Hz). The term RE stands for the relative error between the computed wave velocities by the FEM and the 3L-TB. The term BL stands for the baseline condition or the undamaged condition of the structure (0 s < t < 12 s).

			Acceleration waves				Strain waves					
			F	EM	3L	3L-TB		FI	FEM		3L-TB	
PGA	<i>z</i> [m]	<i>l_i</i> [m]	τ_i [ms]	$v_i \ [m/s]^a$	$\tau_i [\mathrm{ms}]$	$v_i \ [m/s]^a$	RE (%) ^b	τ_i [ms]	$v_i \ [m/s]^a$	τ_i [ms]	$v_i \ [m/s]^a$	RE (%) ^b
BL	25.9 - 41.6	15.7	39.152	400.305	41.031	381.969	-4.58	-	-	-	-	-
	8.8 - 25.9	17.1	37.097	460.708	35.429	482.392	4.71	42.050	779.166	39.509	829.262	6.43
	0.0 - 8.8	8.8	13.650	647.339	13.073	675.910	4.41	29.016	304.536	27.455	321.847	5.68
0.1g	25.9 - 41.6	15.7	39.180	400.020	41.031	381.969	-4.51	-	-	-	-	-
	8.8 - 25.9	17.1	37.264	458.638	36.437	469.057	2.27	42.059	779.000	41.031	798.501	2.50
	0.0 - 8.8	8.8	13.642	647.745	12.593	701.710	8.33	29.022	304.471	27.458	321.819	5.70
0.15g	25.9 - 41.6	15.7	39.188	399.936	41.031	381.969	-4.49	-	-	-	-	-
	8.8 - 25.9	17.1	37.261	458.678	36.946	462.591	0.85	41.557	788.395	40.530	808.374	2.53
	0.0 - 8.8	8.8	13.866	637.251	13.562	651.570	2.25	30.021	294.337	28.474	310.326	5.43
0.2g	25.9 - 41.6	15.7	39.198	399.831	40.530	386.692	-3.29	-	-	-	-	-
	8.8 - 25.9	17.1	37.301	458.190	37.447	456.401	-0.39	41.557	788.395	39.509	829.262	5.18
	0.0 - 8.8	8.8	14.608	604.881	14.052	628.849	3.96	30.517	289.551	31.036	284.715	-1.67

^a $v_i = l_i / \tau_i$

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^b $RE = (v_{i,TB} - v_{i,FEM}) / v_{i,FEM}$



Figure 17: Variation of acceleration wave velocities (a), strain wave velocities (b), and global wave velocities (c) computed by the peak-picking analysis of the IRFs obtained by the FEM of the tower (v_i) with respect to their undamaged or baseline values (BL, v_i^o). (0.8-20 Hz, 0 s < t < 12 s, $F_s = 100$ Hz).

Finally, this section is concluded with the analysis in Fig. 18 on the relationship between the fitted earthquakeinduced stiffness reductions *EI* in Section 4.2.3 and the previously reported wave velocities v_i . In particular, the relative variation of v_i in the bottom layer (L1) computed by the peak-picking analysis of the IRFs obtained by the FEM is plotted against the corresponding *EI* values of the fitted three-layered TB model (3L-TB). In this figure, the velocities obtained for both strain and acceleration waves are presented. Firstly, it is noted that decreases in the bending stiffness yield monotonic decreases in both the strain and acceleration wave velocities. It is also observed that the results obtained for strain waves exhibit a different trend, with decreasing variation rates after PGA=0.15g.

⁵⁹² These results demonstrate that acceleration-based wave propagation approaches are superior, while the strain-free

⁵⁹³ condition at the roof level of cantilever towers limits the accuracy of strain-based approaches.



Figure 18: Relative variation of wave velocities (v_i) in the first layer L1 computed by the peak-picking analysis of the IRFs obtained by the FEM of the tower versus relative variation of bending stiffness (*E1*) computed by the three-layered TB model (3L-TB) with respect to their undamaged or baseline values $(v_i^o \text{ and } EI^o)$. (0.8-20 Hz, 0 s < t < 12 s, $F_s = 100 \text{ Hz}$).

594 5. Conclusions

This paper has proposed the use of deconvolution seismic interferometry for earthquake-induced damage iden-595 tification in historic masonry towers. An analytical multi-layered Timoshenko beam model has been also intro-596 duced for wave propagation analysis and for defining an inverse problem suitable for damage identification. On 597 this basis, detailed parametric analyses have been presented to illustrate the structure of the TB model. In particu-598 lar, both acceleration and strain wave propagation analyses have been performed, and the discussion has focused 599 on the effects of dispersion upon the identification of travelling pulses. A validation case study of a 41.6 m high 600 masonry tower under synthetic earthquake ground motion has been presented. To do so, a non-linear 3D FEM 601 has been built and used to generate pseudo-experimental structural response data under seismic excitation caus-602 ing increasing damage severities. Afterwards, IRFs based on strain and acceleration records have been computed 603 at different heights of the FEM, and the system and damage identification through inverse calibration of the TB 604 model and simple peak-picking analysis have been discussed. On one hand, it has been demonstrated that damage 605 identification can be performed in a fully data-driven way by peak-picking the arrival times of identified pulses 606 in the IRFs of seismic response records. On the other hand, the inverse calibration of the TB model has been 607 also reported to be appropriate for such a purpose, with the added advantage of relating the identified wave delays 608 to earthquake-induced effects on the intrinsic stiffness of the structure. In addition, the TB model has proved to 609 be a suitable tool for identifying not only dominant pulses but the complete waveforms, what circumvents the 610 difficulties associated with the identification of travelling pulses in highly dispersive systems. 611

⁶¹² The main key findings of this work can be summarized as follows:

• The shear-wave propagation problem in a Timoshenko beam is characterized by two different propagation modes. Below a critical frequency $\omega_{cr} = \sqrt{\kappa c_S}/r_g$, the second propagation mode only contributes with evanescent waves. Conversely, for frequencies above this threshold, both modes manifest as propagating waves and the resulting waveform is characterized by complex interference patterns. Thereby, in application to masonry towers, it is necessary to low-pass filter the seismic records with cut-off frequency ω_{cr} to filter out the propagating waves associated with the second propagation mode. Practically speaking, this cut-off frequency is in the order of 46 Hz for towers of slenderness ratio $H/r_g = 17.4$, and moduli ratio R = 0.4.

• The strain-free condition at the roof level of masonry towers precludes the development of strain waves at that level. Hence, the appearance of identifiable strain pulses on the top of the tower requires larger distances compared to the bottom part of the structure, what hinders the applicability of strain-based wave propagation analysis to identify regions of the structure close to the roof level. This limitation does not affect acceleration waves and, therefore, acceleration-based systems are more suited for damage identification in masonry towers.

• Since strain waves are highly conditioned by local stiffness effects, the TB model may fail at representing local wave attenuations. Nevertheless, the presented results have demonstrated that the TB model remains apt for representing the scattering of strain waves. Conversely, acceleration waves are not so influenced by local effects, but rather by the global stiffness between pairs of sensors and, therefore, the TB model yields accurate results for both the attenuation and scattering of acceleration waves.

- The results have shown that the stiffness distribution derived from the inverse calibration of the TB model is well correlated with the earthquake-induced damage and, thence, it offers a valuable tool for damage identification, that is, damage detection, localization, and quantification. This approach identifies the complete wavefront within a certain frequency band and, therefore, there is no need to identify dominant pulses, what may be difficult in highly dispersive systems.
- The results have also reported the possibility of performing seismic interferometry of masonry towers in a fully data-driven way by simple peak-picking. In particular, it has been shown that the measured wave delays are well correlated with the earthquake-induced damages, so that damages can be detected, located and, to some extent, quantified only using seismic response records.

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